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# Evaluation of mechanical and erosive wear properties of hybrid nanocomposites Basalt/E-Glass fiber + MWCNTs/SiO<sub>2</sub>

**Abstract:** The extensive study conducted has revealed that the mechanical and other characteristics of hybrid composites are greatly influenced by several aspects, such as the kind of fibre reinforcement, nanofillers, matrix materials, and the manufacturing methods employed. In the context of structural applications, the incorporation of Multi-wall Carbon Nanotubes (MWCNTs) as fillers in E-Glass fibres had a significant impact on the rapid alteration of mechanical properties, particularly strength, in the composites. This study aimed to create hybrid nanocomposites by including multi-walled carbon nanotubes (MWCNTs) and silicon dioxide (SiO<sub>2</sub>) as filler materials, Basalt/E-glass as fibres, and epoxy with hardener as the matrix. This study utilises statistical methods, specifically the Taguchi L16 Orthogonal Array (OA), to analyse the impact of various operating factors. The parameters selected for this study include MWCNTs/SiO<sub>2</sub> fillers at varying weight percentages (0%, 1%, 2.9%, and 3%), an epoxy matrix with decreasing weight percentages (40%, 39%, 38%, and 37%), compression pressures ranging from 5 MPa to 35 MPa, and moulding temperatures ranging from 40°C to 100°C. The mechanical parameters being examined are the tensile strength and erosive wear. The utilisation of the hand layup and compression moulding technique achieved the development of these composites. By optimising the moulding temperature, significant improvements were achieved in the tensile strength, which reached 184.38 MPa. Similarly, reducing the percentage of fillers resulted in a substantial decrease in erosive wear, with a value of 95.06 mg/kg.

**Keywords:** Hybrid Fillers, Compression Pressure, moulding temperature, Taguchi optimization

## 1. Introduction

Composite materials are replacing ceramics, wood, and metals due to their processing ease and productivity (Pani et al., 2019). Polymeric composites are preferred for their strength and stiffness in maritime, chemical, electrical, automotive, aerospace, and biomedical applications (Shettar et al., 2020). Synthetic or natural fibres in polymeric composites give them unique features for these applications. Carbon, Glass, and Basalt synthetic fibres are appropriate for bearing fabrication since they are stiffer and stronger than MMC (Namdev et al., 2022; Natrayan et al., 2022; Chin et al., 2020). Glass-fiber reinforcement is ideal for aerospace applications due to its mechanical and thermal qualities. Polymer composites with matrix materials like polyurethane, polyester, epoxy,

33 polypropylene, and polyvinyl butyrate are used in ballistic armour, bulletproof vests, and helmets.  
34 Hybrid composites are used in many applications because epoxy resin increases their mechanical  
35 strength (Baby et al., 2019; Hsissou et al., 2021; Jiang et al., 2021). Epoxy matrix materials lower  
36 composite density, making them useful for lightweight component production. For aerospace  
37 applications, glass-fiber epoxy composites have strong adhesive bonding (Choudhary et al., 2019).  
38 Industrially important research into epoxy-based synthetic fibre composite materials' processing  
39 methods for polymer-based composite strength and hardness (Deeban et al., 2023).  
40 Polymer-based epoxy composites, especially those with glass and carbon fibres, have excellent  
41 tribological, mechanical, and physical properties, making them appropriate for various  
42 applications (Singh et al., 2021). Carbon, glass, and other fibres strengthen PMCs, while epoxy,  
43 phenolic, polylactic acid, polyester, and vinyl ester provide toughness, corrosion resistance, and  
44 heat resistance (Kim et al., 2021). Although several matrix materials are used, epoxy resins are  
45 particularly helpful in aerospace, automotive, and marine applications due to their strength. Since  
46 its introduction as a matrix material, epoxy resin has improved in wear resistance in excavator  
47 engine hoods, stiffness and strength in engine frames, energy absorption, tensile, bending, and  
48 impact strength and load carrying capacity, hydraulic cylinder weight reduction, and more  
49 (Velmurugan et al., 2023; Yao et al., 2021; et al., 2021).  
50 Composites are made by compression moulding, vacuum infusion, manual lay-up, spray-up,  
51 vacuum bag moulding, resin transfer moulding, extrusion, injection moulding and filament  
52 winding. (Omran et al., 2021). Every process has pros and cons, but compression moulding's  
53 inexpensive tooling stands out. Compression moulding improves production rates, dimensional  
54 stability, repeatability, surface smoothness, mechanical properties, flame resistance, and more. (Li  
55 et al., 2020; Jose et al., 2022; Dani et al., 2021). Compression moulding procedures for epoxy-  
56 based fibre-reinforced polymer composites need more research, according to the literature. In the  
57 automotive and aerospace industries, compression moulding produces high-strength composite  
58 components (Murasing et al., 2021). This method creates complex-shaped carbon-fiber epoxy  
59 composites with exceptional mechanical properties. High-temperature compression moulded  
60 natural fibre and polylactic acid non-woven mats are also available (Thooyavan et al., 2021).  
61 Taguchi optimising compression moulding parameters like temperature, pressure, and duration  
62 enhanced polylactic acid polymer/woven flax composite tensile strength. Compression moulding

63 has increased the <sup>7</sup>stiffness and strength of carbon fiber-reinforced composites in most aerospace  
64 products (Jeemo et al., 2021).

65 Recently developed glass fibre polymer composites have used various natural fibres to improve  
66 mechanical properties and wear resistance. Reinforcing glass fibre composites using sisal and  
67 wood improves their tensile, hardness, flexural, and impact properties (Singh et al., 2019; Ogbonna  
68 et al., 2021; Bhat et al., 2021). <sup>12</sup>Glass fiber-reinforced epoxy composites with silica nanoparticles  
69 have quintupled fatigue strength. Interlaminar shear, bending, fracture, and impact strengths have  
70 improved significantly with basalt, jute, and flax fibres in glass fibre-reinforced polymer  
71 composites (Alshahrani et al., 2022; Chiang et al., 2020). Natural fibre should weigh less than  
72 glass fibre for hybrid composites to have the best mechanical strength. In hybridization, 10% of  
73 jute and tea leaf fibres produced mechanical strength greater than glass fibre composites.

74 Sisal, kenaf, and jute reinforcement of glass fibre or polymer composites have consistently  
75 improved mechanical properties (Ahmedizat et al., 2019). Hard particles in material erosion cause  
76 wear, one of several forms of material degradation. Polymer composites are used in many  
77 industrial situations; hence they must be worn-tested before use. Several studies have examined  
78 polymer composites' <sup>2</sup>erosive wear using different fibres and fillers. Vacuum-assisted  
79 manufacturing created glass fiber-reinforced polymer composites with micro silica and zinc oxide,  
80 which were tested for erosive wear at various impingement angles. In the study, zinc oxide  
81 increased erosive wear while silica fumes strengthened the composite (Tian et al., 2021). Larger  
82 erodent particles and higher incursion angles increased erosive wear. Marble dust reduces erosive  
83 wear in glass fiber-reinforced polymer composites by enhancing stiffness and hardness (Eayal et  
84 al., 2021). To reduce composite wear, matrix type, filler type, and production methods can be  
85 changed.

86

87 This study focuses on hybrid nanocomposites made with MWCNT and SiO<sub>2</sub> filler materials to  
88 reinforce basalt/E-glass fibres and epoxy with a hardener. These composites are hand-laid and  
89 compressed moulded. The Taguchi technique optimises process parameters for composite  
90 preparation.

91

92

93

## 94 2. Materials and Methods

95 Two filler materials were used in this study. Multi-Walled Carbon Nanotubes (MWCNTs) with  
96 specified characteristics were the first filler. The inner and outer diameters of these MWCNTs are  
97 3 and 6 nm. MWCNTs average 18 mm long and 98% pure. Ultrananotech Private Limited in  
98 Karnataka supplied these MWCNTs. A second filler material used in this investigation is silicon  
99 dioxide. These 30 nm SiO<sub>2</sub> particles are 99.95% pure. SiO<sub>2</sub> filler was also purchased from  
100 Ultrananotech Private Limited in Karnataka, like MWCNTs.

101

102

23  
Table 1. Chemical composition of the Basalt fiber and an E-glass fiber

Fiber Element	FeO/Fe <sub>2</sub> O <sub>3</sub> [wt%]	CaO [wt%]	Na <sub>2</sub> O [wt%]	SiO <sub>2</sub> [wt%]	K <sub>2</sub> O [wt%]	Al <sub>2</sub> O <sub>3</sub> [wt%]	MgO [wt%]	TiO <sub>2</sub> [wt%]
Basalt	8.6	7.6	2.8	50.6	2.6	19.5	4.7	2.2
E - Glass	0.28	20.03	0.40	58.20	0.41	11.78	0.52	0.38

103

104 Basalt and E-Glass fabric fibres were purchased in sufficient amounts for this study's  
105 structural components. Table 1 shows their chemical compositions. Chennai-based HAYAEL  
106 Aerospace Pvt Ltd supplied these fibres. Araldite LY 556 was the basic resin and Aradur HY 951  
107 the hardener in the epoxy matrix composite production process. Bisphenol-A-derived Araldite LY  
108 556 is medium-viscosity, unaltered liquid resin. Aradur HY 951 is a low-viscosity, unaltered  
109 aliphatic polyamine. These ingredients provide a solvent-free, low-viscosity system that cures at  
110 room temperature. This system's reactivity can be changed to meet varied processing and curing  
111 circumstances by adjusting Aradur HY 951 concentration from 10 to 12 parts. Hand layup and  
112 compression moulding were used to make nanocomposites.

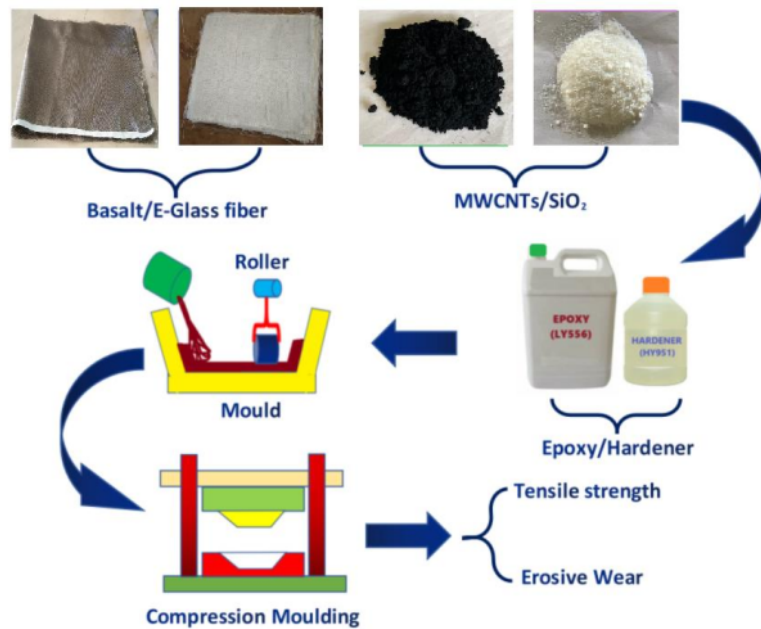
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## 114 3. Experimental Procedure

115 Deionized water was used to thoroughly clean MWCNTs and SiO<sub>2</sub> surfaces. MWCNTs and SiO<sub>2</sub>  
116 were then combined with epoxy resin at 40%, 39%, 38%, and 37% weight percentages.  
117 Ultrasonication with a Q700 sonicator @ 30 Hz for 4 hours followed mixing. For homogenous  
118 nanoparticle dispersion, epoxy, MWCNTs, and SiO<sub>2</sub> were mixed with a hardener using a magnetic  
119 stirrer. The epoxy-to-hardener mixing ratio was 10:1, using 350g epoxy and 35g hardener.  
120 Composite samples were made with Basalt/E-Glass fibres and several hybrid combinations



121 (BGFR-E1, BGFR-E2, BGFR-E3, BGFR-E4) and MWCNTs/SiO<sub>2</sub> weight percentages. Figure 1  
122 shows experimental details.



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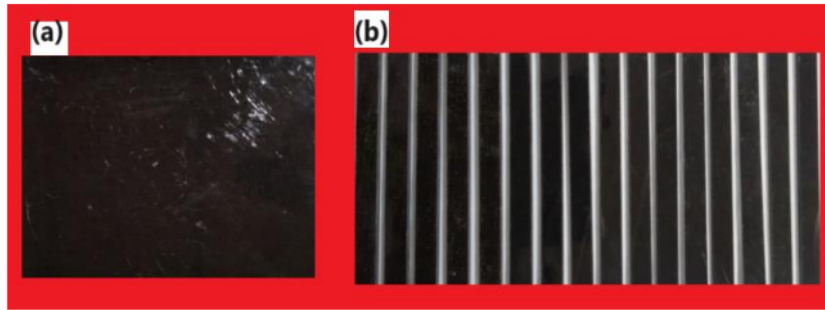
**Figure 1. Experimental flow chart**

125 **3.1** Fibre reinforcing ratio of Basalt/E-Glass maintained 60%. A new 12-layer stacking sequence  
126 of laminates using 6 layers of Basalt fibre and 6 layers of E-Glass fibre created multiple layers of  
127 composite materials. The epoxy-filler mixture was mixed and hand-laid to make these laminated  
128 composites. Epoxy-filler solutions were put between 12 alternate layers of Basalt/E-Glass fabric.  
129 A flat roller compressed laminate surfaces to produce uniform stack thickness. The assembled  
130 laminates in the mould were then compressed at various pressures (5 MPa, 15 MPa, 25 MPa, and  
131 35 MPa) and temperatures (40°C, 60°C, 80°C, and 100°C). Once curing was complete, 200 mm x  
132 200 mm × 6 mm sheets were taken from the mould for testing.

133

### 134 **3.2 Tensile test**

135 A Universal Testing Machine (UTM) from Shimadzu Corporation in Kyoto, Japan, was used to  
136 test composites' tensile strength. Tensile testing specimens were prepared according to ASTM D  
137 638-03. Each test was conducted numerous times, using 30 × 30 × 5 mm samples. Figure 2  
138 shows the raw sheet and tensile test specimens.



139

140 **Figure 2 Basalt/E-Glass fiber/ MWCNTs/SiO<sub>2</sub> Hybrid nanocomposite: (a) Fabricated Panel**

141 **(b) Tensile test specimens**

142 **3.2 Erosive Wear Analysis**

143 The erosion wear test was conducted according to ASTM-G 76 standards, using specimens  
 144 measuring 30 × 30 × 5 mm<sup>3</sup>. The investigation utilized an air jet erosion tester supplied by  
 145 DUCOM.

146 **Table 2 Independent variables and dependent responses**

Exp. Run s	Filler s (%)	Epoxy matrix (%)	Compression pressure (MPa)	Moulding temperature (°C)	Tensile Strength (MPa)	Erosive Wear (mg/kg)	Predicted Tensile strength (MPa)	Predicted Erosive Wear (mg/kg)
1	0	40	5	40	164.02	184.37	163.57	184.17
2	0	39	15	60	172.96	103.88	173.21	101.1
3	0	38	25	80	175.08	103.47	174.84	102.96
4	0	37	35	100	182.98	95.06	183.42	98.54
5	1	40	15	80	173.87	224.16	174.31	227.64
6	1	39	5	100	180.18	145.78	179.94	145.27
7	1	38	35	40	170.46	194.85	170.71	192.07
8	1	37	25	60	178.68	196.64	178.23	196.44
9	2	40	25	100	181.28	347.05	181.53	344.27
10	2	39	35	80	180.4	302.27	179.95	302.07
11	2	38	5	60	175.52	192.4	175.96	195.88
12	2	37	15	40	172.61	271.66	172.37	271.15
13	3	40	35	60	178.77	413.06	178.53	412.55
14	3	39	25	40	171.55	406.38	171.99	409.86
15	3	38	15	100	184.38	249.45	183.93	249.25
16	3	37	5	80	179.98	297.26	180.23	294.48

147 For this study, various erodent was considered, with a specific focus on silica particles  
148 measuring 150 nm in size. The impingement force was applied to the specimen surfaces for a  
149 duration of 20 minutes, achieved through a tungsten carbide nozzle. The selected parameters for  
150 this study included an impact velocity of 40 m/s and an impingement angle of 45 degrees, with  
151 different experimental conditions. Each sample was weighed both before and after the test to  
152 calculate the weight loss.

153 This investigation employed the Taguchi optimization technique to assess the tensile  
154 strength and erosive wear characteristics of laminated composites under various operational  
155 conditions. Table 2 furnishes the process parameters, their corresponding levels and responses for  
156 all sixteen experimental runs.

157

#### 158 **4. Results and Discussion**

159 Table 2 provides a comprehensive summary of the analysis conducted on tensile and  
160 erosive wear, comprising sixteen experimental runs. The highest recorded tensile strength was  
161 184.38 MPa, which occurred when 3% fillers were present, along with a 38% epoxy matrix, a  
162 compression pressure of 15 MPa, and a moulding temperature of 100°C. Conversely, the absence  
163 of fillers, coupled with maximum compression pressure and moulding temperature, resulted in  
164 superior wear resistance like 95.38 mg/kg. As both the key properties are important here they are  
165 discussed comparatively.

166 In the tensile strength analysis of tensile strength, control factor values were converted into  
167 mean and S/N ratio values, effectively illustrating the factors influencing tensile strength. Table 3  
168 displays the mean and S/N ratio values for the tensile strength analysis, revealing that moulding  
169 temperature had the most significant impact, followed by fillers percentage, epoxy matrix, and  
170 compression pressure. The optimized parameters for achieving maximum tensile strength were  
171 identified as follows: 3 % fillers, 40 % epoxy matrix, 35 MPa compression pressure, and a  
172 moulding temperature of 100°C.

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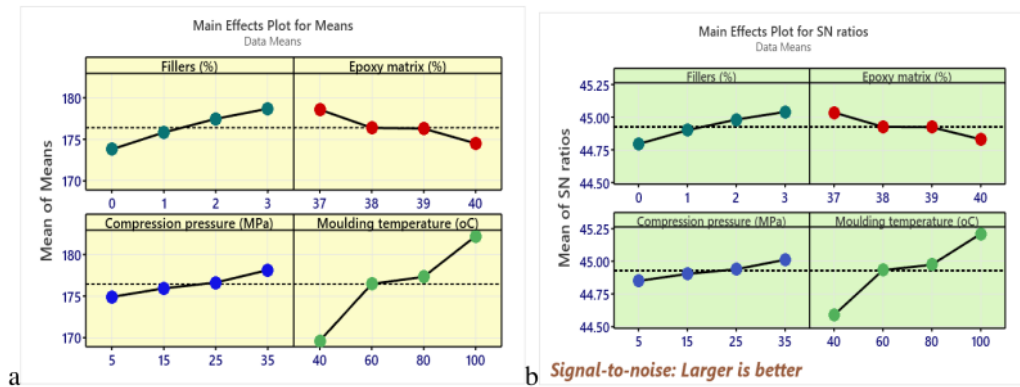
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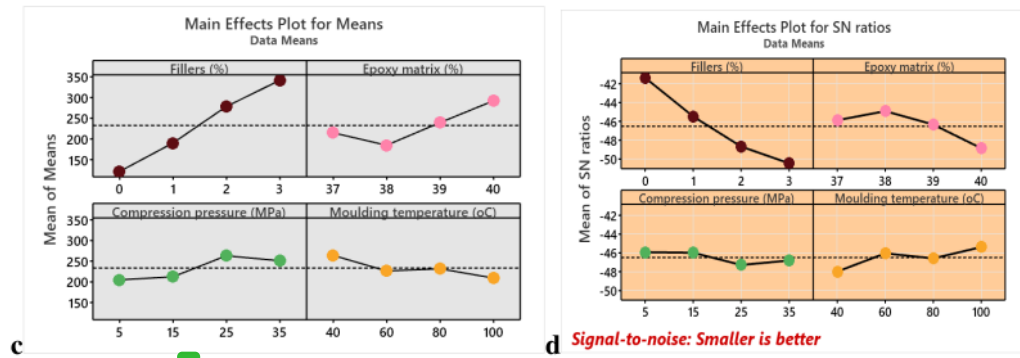
**Table 3. Taguchi Results for the Responses of tensile strength and erosive wear**

Level	Means of Tensile Strength				S/N ratio for Tensile Strength			
	Fillers (%)	Epoxy matrix (%)	Compression pressure (MPa)	Moulding temperature (°C)	Fillers (%)	Epoxy matrix (%)	Compression pressure (MPa)	Moulding temperature (°C)
1	173.8	178.6	174.9	169.7	44.79	45.03	44.85	44.59
2	175.8	176.4	176	176.5	44.9	44.92	44.9	44.93
3	177.5	176.3	176.6	177.3	44.98	44.92	44.94	44.97
4	178.7	174.5	178.2	182.2	45.04	44.83	45.01	45.21
Delta	4.9	4.1	3.2	12.5	0.25	0.21	0.16	0.62
Rank	2	3	4	1	2	3	4	1
Criteria	<b>Larger is better for Tensile Strength</b>							
Level	Means of erosive wear				S/N ratio for erosive wear			
1	121.7	215.2	205	264.3	-41.38	-45.89	-45.93	-47.99
2	190.4	185	212.3	226.5	-45.49	-44.93	-45.99	-46.05
3	278.3	239.6	263.4	231.8	-48.7	-46.35	-47.29	-46.59
4	341.5	292.2	251.3	209.3	-50.48	-48.86	-46.82	-45.4
Delta	219.8	107.1	58.4	55	9.1	3.93	1.36	2.6
Rank	1	2	3	4	1	2	4	3
Criteria	<b>Smaller is better for erosive wear</b>							

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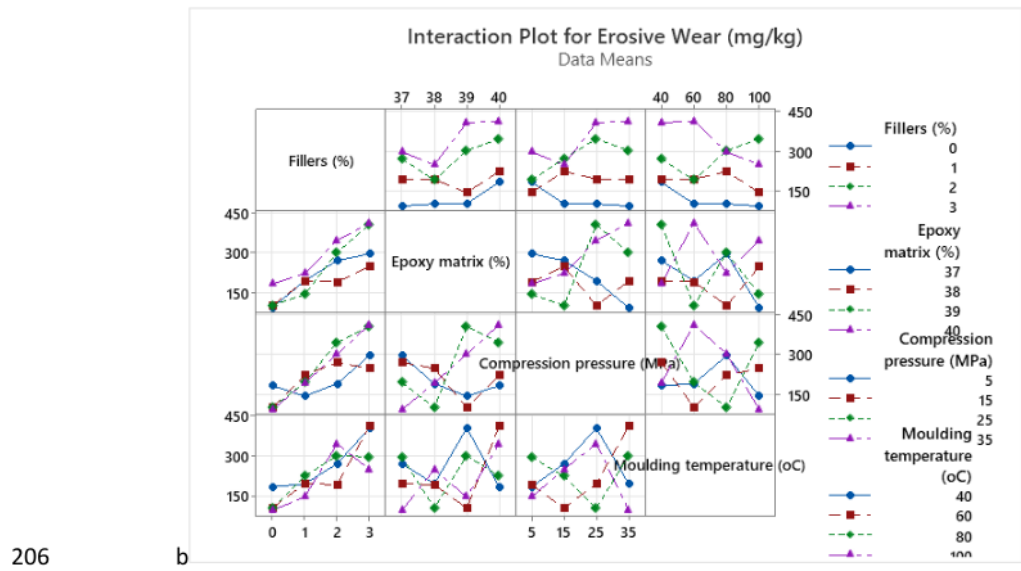
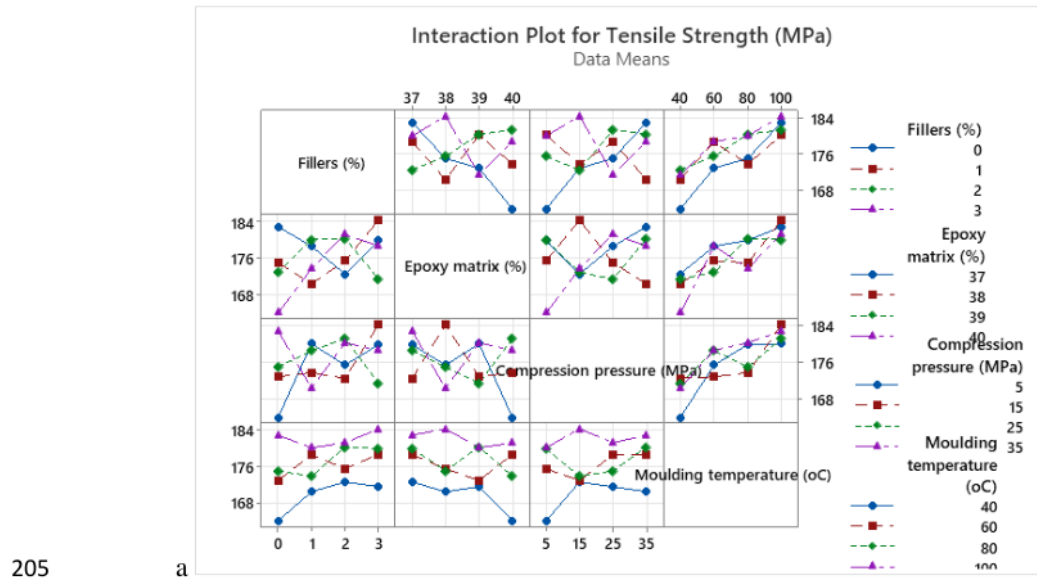
184 **Figure 2 mean and S/N ratio plots for the responses of tensile strength and erosive wear**

185

186 In the assessment of erosive wear, a "smaller is better" criterion was adopted, and all  
 187 parameter values were converted into mean and S/N ratio values. Table 6 illustrates the mean and  
 188 S/N ratio values for the erosive wear analysis. Among all the parameters examined, the filler  
 189 percentage exhibited the most substantial influence, followed by the epoxy matrix, moulding  
 190 temperature, and compression pressure. Ultimately, the optimal parameters identified for the  
 191 erosive wear analysis were as follows: 0% fillers, 38% epoxy matrix, 5 MPa compression pressure,  
 192 and a moulding temperature of 100°C.

193 Figure 2a and Figure 2b depict main effect plots for both mean values and Signal-to-Noise  
 194 (S/N) ratio. It is evident from the figure that as the filler percentage increases (reaching 3%), there  
 195 is a notable enhancement in tensile strength. This trend is also observed with compression  
 196 moulding pressure at 35 MPa and moulding temperature at 100°C. In contrast, when it comes to  
 197 the epoxy matrix, the lowest level (37%) exhibits the highest tensile strength. As the percentage  
 198 of epoxy matrix increases beyond this point, there is a decrease in tensile strength observed. Figure

199 2c and Figure 2d illustrate an interaction plot that effectively depicts the impact of various  
 200 parameters on erosive wear. The plot unmistakably demonstrates that superior erosive wear  
 201 performance is associated with lower values of the parameters. Notably, in this analysis, when no  
 202 filler material is added, reducing the epoxy matrix content, decreasing compression moulding  
 203 pressure, and increasing moulding temperature all lead to a marked reduction in wear rates,  
 204 signifying a significant enhancement in erosive wear resistance.



207 **Figure 3. Interaction plot (a) for tensile strength (b) for Erosive wear**

208 In Figure 3a, the interaction plot displays the relationship between tensile strength and  
209 various parameters. It is evident from the plot that higher values of these parameters correspond to  
210 superior tensile strength. Specifically, as the percentage of filler material, compression moulding  
211 pressure, and moulding temperature increase, there is a noticeable improvement in tensile strength.  
212 Figure 3b illustrates an interaction plot that effectively depicts the impact of various parameters  
213 on erosive wear. The plot unmistakably demonstrates that superior erosive wear performance is  
214 associated with lower values of the parameters. Notably, in this analysis, when no filler material  
215 is added, reducing the epoxy matrix content, decreasing compression moulding pressure, and  
216 increasing moulding temperature all lead to a marked reduction in wear rates, signifying a  
217 significant enhancement in erosive wear resistance.

218 The statistical significance of the control factors in the tensile strength analysis was  
219 assessed using Analysis of Variance (ANOVA) at a 95% confidence level, and the results are  
220 presented in Table 4. In this table, the p-values for all process parameters were clearly reported.  
221 Notably, all of these p-values were found to be less than 0.05, indicating a high level of  
222 significance. This compelling evidence from the ANOVA results underscores that the selected  
223 parameters and their respective levels have a substantial influence on the tensile strength.  
224 Furthermore, when examining the percentage of contribution to the tensile strength, it becomes  
225 evident that the moulding temperature played the most significant role, accounting for 68.59% of  
226 the variation. Following closely behind was the filler percentage at 12.43%, the epoxy matrix at  
227 7.03%, and the compression pressure at 4.98%. In the case of erosive wear, Table 4 presents the  
228 results of the Analysis of Variance (ANOVA) conducted to assess the statistical significance of  
229 the control factors in the analysis of erosive wear. This analysis was performed at a 95%  
230 confidence level. The table includes F-values for all process parameters, which were used to  
231 determine the contribution percentage of each parameter. Importantly, all the p-values for these  
232 parameters were found to be less than 0.05, indicating a high level of significance. The ANOVA  
233 results provide strong evidence that the selected parameters and their respective levels significantly  
234 influence erosive wear. Upon closer examination of the percentage of contribution to the erosive  
235 wear analysis, it becomes clear that the Moulding temperature ( $^{\circ}\text{C}$ ) has the highest influence  
236 ( $p=0.001$ ) followed by the filler's percentage ( $p=0.012$ ), epoxy matrix ( $p=0.023$ ) and compression  
237 pressure ( $p=0.042$ ). The prediction model was developed and presented to predict the tensile  
238 strength and erosive wear in Equation 1 and Equation 2 respectively.

239

240

5 **Table 4. Results of Analysis of Variance for Tensile Strength and Erosive Wear** 16

	Source	DF	Adj SS	Adj MS	F-Value	P-Value
Tensile Strength	Regression	4	401.80	100.449	36.72	0.000
	Fillers (%)	1	53.69	53.694	19.63	0.001
	Epoxy matrix (%)	1	30.36	30.356	11.10	0.007
	Compression pressure (MPa)	1	21.53	21.528	7.87	0.017
	Moulding temperature (°C)	1	296.22	296.219	108.28	0.000
	Error	11	30.09	2.736		
	Total	15				
Erosive Wear	Fillers (%)	3	54.367	18.122	26.33	0.012
	Epoxy matrix (%)	3	33.440	11.147	16.20	0.023
	Compression pressure (MPa)	3	22.018	7.339	10.66	0.042
	Moulding temperature (°C)	3	320.002	106.667	154.98	0.001
	Error	3	2.065	0.688		
	Total	15	431.891			

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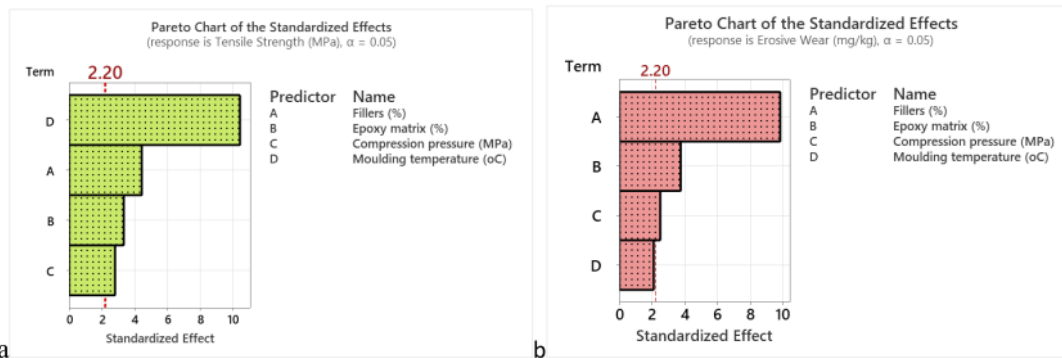
243 Regression Equation

244 Tensile Strength (MPa) = 205.8 + 1.639 Fillers (%) - 1.232 Epoxy matrix (%) + 0.1037  
 245 compression pressure (MPa) + 0.1924 Moulding temperature (°C) (1)

246

247 Erosive Wear (mg/kg) = -961 + 74.75 Fillers (%) + 28.56 Epoxy matrix (%) + 1.902 Compression  
 248 pressure (MPa) - 0.798 Moulding temperature (°C) (2)

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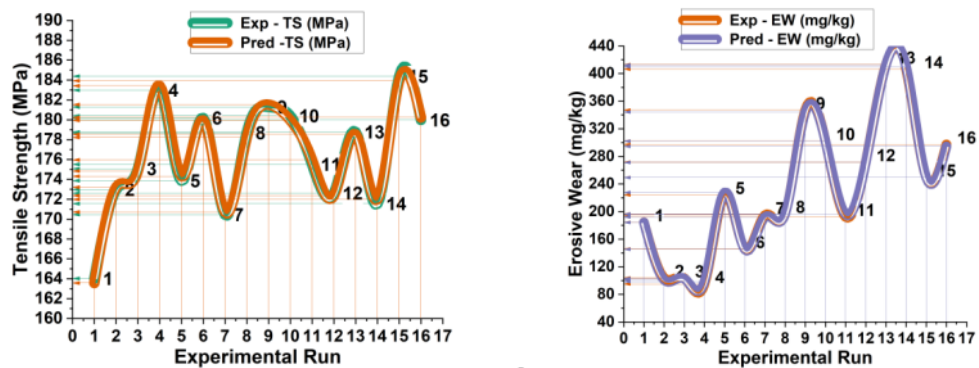
**Figure 4. Pareto analysis (a) for Tensile strength (b) for Erosive wear**



252 In Figure 4a, all parameter values have exceeded the reference line. The moulding temperature  
 253 for D has surpassed the maximum level, followed by Fillers percentage for A, Epoxy matrix  
 254 percentage for B, and Compression pressure for C. Figure 4b displays a Pareto chart that  
 255 facilitates the analysis of parameter contributions in the erosive wear analysis. With the exception  
 256 of the moulding temperature (D), all other parameter values have surpassed the reference line.  
 257 Notably, the filler percentage for parameter A clearly exceeds the maximum level indicated by  
 258 the reference line when compared to the other parameters. Following this, parameter B,  
 259 representing the epoxy matrix, and parameter C, corresponding to compression pressure, also  
 260 exhibit values that exceed the reference line.

261 Figure 5a illustrates a comparative analysis between the experimental and predicted tensile  
 262 strength values, revealing that the majority of the experimental results closely matched the  
 263 predicted values. This suggests that the tensile tests were effectively executed, and the influence  
 264 of each process parameter was evident throughout all the experimental runs. Figure 5b illustrates  
 265 a comparative analysis between the experimental and predicted erosive wear data. The analysis  
 266 demonstrates that a significant portion of the experimental results closely align with the predicted  
 267 values. This indicates that the erosive wear results were obtained with a high degree of accuracy,  
 268 reflecting the effectiveness of the test execution. Furthermore, it is evident that each process  
 269 parameter had a discernible impact on the outcomes across all experimental runs.

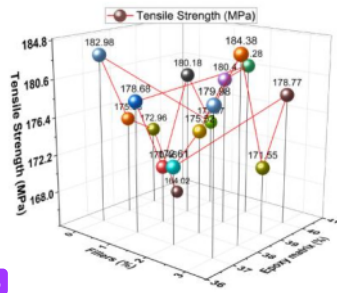
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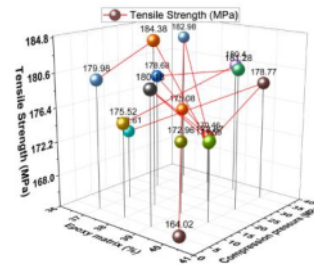
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272 **Figure 5 Run-wise Experimental & Predicted (a) tensile strength (b) Erosive wear**

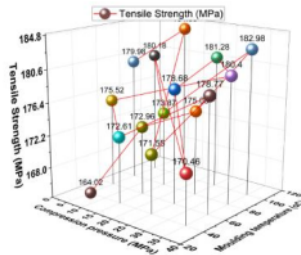


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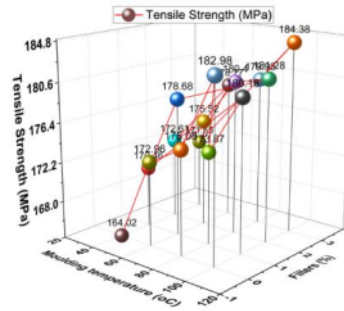


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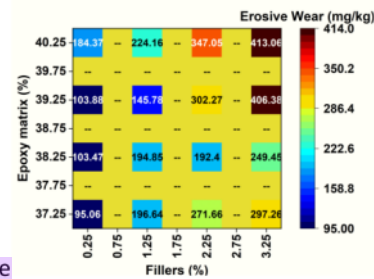


c

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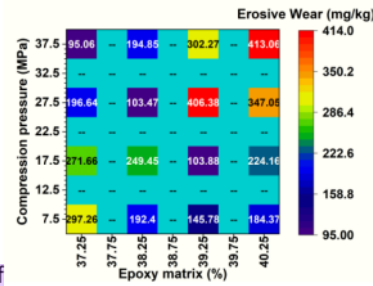


d

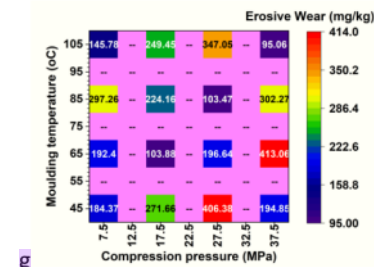


e

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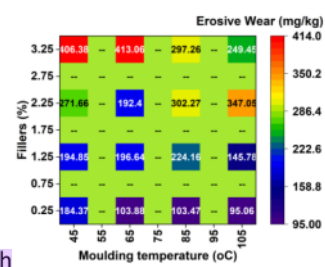


f



g

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h

277

278 **Figure 6. Factors Influencing analysis for Tensile strength (a) Fillers (%) Vs Epoxy matrix**  
 279 **(%), (b) Epoxy matrix (%) Vs compression pressure, (c) Compression pressure Vs moulding**  
 280 **temperature, (d) Moulding temperature Vs Filler (%) for Erosive wear analysis : (e) Fillers**  
 281 **(%) Vs Epoxy matrix (%), (f) Epoxy matrix (%) Vs compression pressure, (g) Compression**  
 282 **pressure Vs Moulding temperature, (h) Moulding temperature Vs Filler (%)**

283

284 Figure 6 presents a 3D trajectory plot that visually depicts the influence of two key  
285 parameters, namely filler percentage and epoxy matrix percentage, on the analysis of tensile  
286 strength. In Figure 6(a), it is evident that the highest levels of filler percentage and epoxy matrix  
287 percentage contribute to greater tensile strength. Conversely, Figure 6(b) demonstrates that the  
288 lowest level of epoxy matrix percentage, coupled with the minimum compression pressure, results  
289 in an increase in the tensile strength of the composites. Figure 6(c) shows that the combination of  
290 minimal compression pressure and the highest moulding temperature leads to a significantly  
291 elevated tensile strength. Finally, Figure 6(d) illustrates that when both filler percentage and  
292 moulding temperature are at their maximum values, the composites exhibit outstanding tensile  
293 strength. Figure 6 e to Figure 6 h provides a visual representation of a heat map plot that illustrates  
294 the impact of two critical parameters in the analysis of erosive wear. In Figure 6 (e), it is evident  
295 that the composites exhibited excellent wear resistance when a minimal filler percentage and a  
296 higher epoxy matrix percentage were employed. Conversely, Figure 6 (f) demonstrates that a lower  
297 epoxy matrix percentage and elevated compression values led to increased wear resistance. Figure  
298 6 (g) highlights that both compression pressure and moulding temperature were conducive to  
299 achieving good wear resistance. Furthermore, Figure 6 (h) reveals that a higher moulding  
300 temperature, in the absence of fillers, resulted in outstanding wear resistance for the composites.

### 301 **Conclusion**

302 Hybrid nanocomposites were successfully fabricated using Basalt/E-Glass fabric fibers,  
303 MWCNTs/SiO<sub>2</sub> fillers, and an epoxy matrix via hand layup and compression moulding techniques.  
304 The study yielded the following key findings:

#### 305 **Tensile Strength Analysis:**

- 306 ❖ The most significant factor impacting tensile strength was the moulding temperature, followed  
307 by the filler percentage, epoxy matrix content, and compression pressure.
- 308 ❖ Optimal conditions for achieving maximum tensile strength were determined as follows: 3%  
309 fillers, 40% epoxy matrix, 35 MPa compression pressure, and a moulding temperature of  
310 100°C.
- 311 ❖ The ANOVA results highlighted the dominant role of moulding temperature, contributing to  
312 68.59%, followed by filler percentage at 12.43%, epoxy matrix content at 7.03%, and  
313 compression pressure at 4.98%.

314 **Erosive Wear Analysis:**

- 315 ❖ Filler percentage exerted the most significant influence on erosive wear, followed by the  
316 epoxy matrix, moulding temperature, and compression pressure.
- 317 ❖ Optimal parameters for minimizing erosive wear were determined as follows: 0% fillers, 38%  
318 epoxy matrix, 5 MPa compression pressure, and a moulding temperature of 100°C.
- 319 ❖ In the erosive wear analysis, the highest impact was attributed to the fillers percentage at  
320 72.98%, followed by the epoxy matrix at 10.65%, compression pressure at 4.72%, and  
321 moulding temperature at 3.33%.
- 322 ❖ Overall, the investigation yielded promising results with a maximum tensile strength of 184.38  
323 MPa and a minimal erosive wear rate of 95.06 mg/kg.

324

325 **Acknowledgement**

326

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329

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